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Template-based geometric transformations of a functionally enriched DMU into FE assembly models

Flavien. Boussuge^{1,2}, Ahmad. Shahwan², Jean-Claude. Léon², Stefanie. Hahmann², Gilles. Foucault² and Lionel. Fine¹

¹EADS – Innovation Works, flavien.boussuge@eads.net, lionel.fine@eads.net

²Grenoble University/INRIA, ahmad.shahwan@grenoble-inp.fr, jean-claude.leon@grenoble-inp.fr, stefanie.hahmann@inria.fr, gilles.foucault@ujf-grenoble.fr

ABSTRACT

Pre-processing of CAD models derived from Digital Mock-Ups (DMUs) into finite element (FE) models is usually completed after many tedious tasks of model preparation and shape transformations. It is highly valuable for simulation engineers to automate time-consuming sequences of assembly preparation processes. Here, it is proposed to use an enriched DMU with geometric interfaces between components (contacts and interferences) and functional properties. Then, the key concept of template-based transformation can connect to assembly functions to locate consistent sets of components in the DMU. Subsequently, sets of shape transformations feed the template content to adapt components to FE requirements. To precisely monitor the friction areas and the mesh around bolts, the template creates sub-domains into their tightened components and preserves the consistency of geometric interfaces for the mesh generation purposes. From a user-selected assembly function, the method is able to robustly identify, locate and transform groups of components while preserving the consistency of the assembly needed for FE models. To enlarge the scope of the template in the assembly function taxonomy, it is shown how the concept of dependent function enforces the geometric and functional consistency of the transformed assembly. To demonstrate the proposed approach, a business oriented prototype processes bolted junctions of aeronautical structures.

Keywords: Assembly, DMU, CAD – CAE integration, Functional information.

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1 NOTATIONS - ACRONYMS

General purpose acronyms:

BC: boundary condition;

DMU: digital mock-up;

FE: finite elements; FEA: finite element analysis(es);

KBE: Knowledge-based engineering;

PDP: product development process;

OWL: web ontology language;

RDF: resource description framework;

STEP: Standard for the Exchange of Product model data;

Acronyms specific to the paper:

CI: conventional interface; CIG: conventional interface graph;

C_c : compatibility conditions between T and S_r ;

FD: functional designation of a component;

FI: functional interface;

I_c : set of components such that each of its components has all its FIs in T;

P_c : set of components such that each of its components has some of its FIs in T;

\mathcal{T}_{ci} : taxonomy of conventional interfaces;

\mathcal{T}_{fd} : taxonomy of functional designations of components;

\mathcal{T}_{fi} : taxonomy of functional interfaces;

\mathcal{T}_{fn} : taxonomy of functions;

S_{me} : reference state expressing static mechanical equilibrium of a component;

S_{lc} : reference state expressing a static load cycle inside an assembly;

S_t : shape transformation incorporated in a template, hence $S_t(T)$; S_{t1} : removal of a threaded link; S_{t2} : cylindrical sub domain around a screw shaft to localize friction effects; S_{t3} : removal of the locking nut of a bolted junction; S_{t4} : screw head transformations; S_{t5} : preserve the cylindrical loose fit between screw and plates to set up contact friction BCs;

T: function-based template performing shape transformations;

T_I : union of all components belonging to I_c .

2 INTRODUCTION

To speed up a PDP, aeronautical, automotive and other companies face increasing needs in setting up FE simulations of large sub-structures of their products. The challenge is to study not only standalone components but to simulate the structural behavior of large assembly structures containing up to thousands of components [9]. DMUs are widely used during a PDP as the virtual geometric product reference. This model contains detailed 3D representation of the whole product structure available for simulation engineers. To prepare large sub-structure models for simulation (such as wings or aircraft fuselage section); the DMU offers a complete geometric input model. However, speeding up the simulation model generation strongly relies on the time required to perform the geometric transformations needed to adapt the DMU to FE requirements.

Currently, time and human resources involved in pre-processing CAD models derived from DMUs into FE models prevent engineers from setting up structural analyses. Very tedious tasks are required to process the large amount of DMU components and the connections between them. The shape transformations required for each assembly component, and interactively performed by engineers, can cover defeaturing [16, 37], idealization [5, 6], domain decomposition [30, 32] as well as the definition of contact areas. Even on repetitive configurations such as junctions, similar geometric transformations must be repeated on all the components and interfaces of all the assembly junctions. Aeronautical structures are particularly complex to transform due to their many independent connectors such as bolts or rivets (see Fig.1). Even though CAE software possess specific operators for bolted junctions, pre-stressed configurations, ..., to speed up FEA pre-processing, the user must already have available a DMU incorporated all the desired geometric transformations. Automated domain decomposition, geometric simplifications, selection of target components are examples of processes still interactive and error prone that a user must perform tediously. Within the available resources and time scale planned in an industrial PDP, the analyst is therefore limited to simulate small models rather than complete assembly structures.

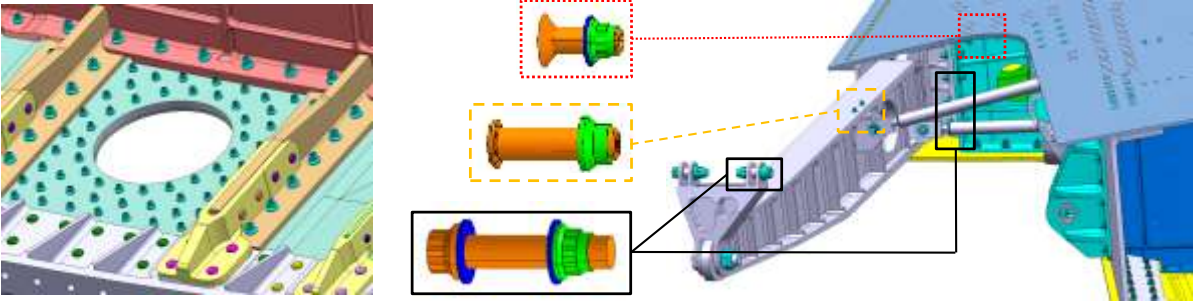


Fig. 1: Examples of aeronautical DMUs with a variety of bolted junctions.

Compared to standalone components, the above observations highlight the prominence of geometric interfaces processing when preparing an assembly model for FEA. Their mechanical behavior and the new shape transformations they convey are specific features of assemblies [9]. As described in [36], component's geometric interfaces are the basis of component's functional designation. Effectively, to speed up the transformation of a DMU into a FE model, the engineer would process the components in accordance to their mechanical effect in the assembly, i.e. their function. Consequently, the relationship between FE simulation hypotheses, component's functions and component's interfaces appears as a key issue to improve the efficiency of DMU transformation for FEA. Indeed, setting up a FEA aims at studying the mechanical behavior of an assembly. Function, behavior and shape share strong connections, as studied by Gero [18, 19] and Albers [1, 2] among others, hence the aforementioned example of the FE preparation process contributing to the representation of bolted connections in an aeronautical assembly.

In this paper, we propose an approach to reduce the FE model preparation time to adapt CAD assembly models derived from DMUs, into FE assembly models. Based on the analysis of DMU transformation requirements for FE assembly model preparation [9], the method uses an enriched DMU as input. It is obtained from previous developments and contains explicit geometric interfaces between components (contacts and interferences) as well as their functional designations. This last step is achieved using a qualitative reasoning process. From this enriched model, it is shown that further enrichment is needed to reach a level of product functions where simulation objectives can be used to specify new geometric operators that can be robustly applied to automate components' and interfaces' shape transformations during an assembly preparation process for FEA. If close to KBE, this scheme is nonetheless more generic and more robust than KBE approaches because functional designations and functions are generic concepts. KBE aims at structuring engineering knowledge and processing it with symbolic representations [14, 27] using language based approaches. Here, the focus is on a robust connection between geometric models and symbolic representations featuring functions.

To prove the validity of our approach and the methodology proposed in [4-9], this paper presents a template-based operator dedicated to automate shape transformations of bolted junctions. Using the functional information and geometric interfaces, the operator applies a user-defined template to simplify bolts and sets control sub-domains around them in their associated tightened components to enable the description of the friction phenomenon between these components. This template is used to precisely monitor the meshing process while preserving the consistency of contacts and adapting the assembly model to simulation objectives.

The rest of the paper is organized as follows. In Section 2, we review the CAD-CAE integration research for FE assembly models. Section 3 reviews the method to enrich the DMU with geometric interfaces between components and functional designations of components. New requirements are introduced to connect functional designations with functions. Section 4, describes the concept of template-based transformation operator and illustrates it with a template-based operator of bolted junctions transformation. Finally, the paper presents the result of this operator on an industrial aeronautical use-case and section 5 exemplifies an extension of the template definition with idealization transformations.

3 RELATED WORK

Assembly models have been studied throughout the development of solid modeling concepts. Essentially, assembly models have been proposed for design and manufacture applications [31-34] with recent applications to collaborative design. With the development of features, approaches set assemblies as components related to each other through geometric constraints [40]. These approaches share a common denominator where assemblies are described as geometric models progressively enriched with technological and then, functional data [15]. However, the closer the functional information, the higher is the requirement to obtain external information to CAD environments and the greater is the need of user's interactions during a design process. Though these additional informations are mandatory in a design process, it does not seem efficient to incorporate them into a preparation process for FE simulation when time reduction is critical to insert simulations into a PDP. If these informations were available at the level of FE simulation, a dedicated approach could be set up to process them but companies are currently not able to set up a digital representation of these data that is connected to the DMUs they produce with CAD systems. Within this framework, Roy and Bharadwaj [33] set up a design approach to connect functions to 3D geometry using a Part Function Model (PFM). There, they address the relationships between function, behavior and geometry of a part in a top-down manner from function to geometry to obtain parts from functional specifications. The PFM described requires up to low level functions that connect part boundary faces to function since the behavior model builds up on interfaces between parts, i.e. contact surfaces to position parts with respect to each other. At the level of complex assemblies with hundreds to thousands of parts, adding the amount of complementary data defining such a behavioral model becomes too tedious. Recent approaches [25] reduce the description of junctions between components to global parameters describing bolted and riveted connections without referring to the individual surfaces of each bolt or rivet. This fits well with a design process but simulation objectives may require a detailed representation of interfaces when the purpose is to assess the stress field distribution in a bolted connection with tens or hundreds of bolts.

Developments of ontology-based approaches take advantage of new capabilities to structure concepts and to connect them with component models [26, 31] or at the level of 3D geometry entities [7-20]. Some of these approaches have been applied to assemblies [7, 23, 31] and can take advantage of reasoners to set up inference rules to ensure the consistency of the assembly description or extract information that is not readily available in the dataset describing an assembly. In [23, 31], there is no direct connection between the ontology content and the 3D model of components. Here, ontologies support a design process where the engineer gets consistent information in a collaborative context but the ontology does not connect to boundary entities of component geometric models. Barbeau et al. [7] cover a larger description of a product with an ontological description incorporating the geometry and structure levels as available through STEP APIs [21, 22], up to the functional description that can be inserted through ontological representation of the Core Product Model (CPM) [17]. The authors showed that not all concepts of STEP could be rigorously expressed using OWL standard [29] leading to limitations in detecting inconsistencies. If CPM is able to relate component functions to form features, the content of the CPM reflects the designers' choices to determine the level of product features inserted in this model and its level of connection with each component's geometry. Consequently, there is no guarantee that functional information can be available downward to the level of B-Rep entities of each component. Reasoning capabilities associated with ontologies are set up to browse the whole assembly ontology but they are not applied to derive new functional information or reinforce the connection between function and geometry in an assembly model. Consequently, we can observe that assembly processing is often addressed in a design context of top-down approach from primary product function point of view and there is no low level robust connection between components' geometry and low level functions. Additionally, the reasoning capabilities of ontology-based approaches are exploited to browse existing assembly data and derive data not readily available. These reasoning processes do not extend further than relative positioning constraints between components. KBE approaches [14, 27] formalize engineering knowledge to automate some design tasks. KBE concepts take advantage of artificial intelligence techniques and strongly rely on language-based approaches. As such, they follow top-down approaches based on enterprise 'best practices' and address preliminary design stages rather than embodiment or detailed design ones, which is the

product geometric description addressed here with DMUs. This is hardly applicable to other companies and even more to other products. Still, if KBE approaches incorporate connections with geometric models, geometry is perceived as attribute of higher level concepts like components, which does not bring new structure to the geometric models of components. Recently, Shahwan et al. [36] inserted a qualitative reasoning process that can relate geometric interfaces up to the functional designation of components, thus creating a robust and automated connection between 3D geometric entities and functional designations of components. It is a bottom-up approach that fits to our current requirements.

Research in CAD-CAE integration has mainly focused on the use of global geometric transformations of standalone CAD components [37]. Few contributions addressed the automation of assembly pre-processing. Automatic simplifications of assembly for collaborative environment as the multi-resolution approach of Kim [24] or surface simplification of Andujar [3] transform assembly components independently of each other. This is not sufficient to pre-process FE assembly models because mechanical joints and interfaces tightening the different components must be part of their pre-processing. Commercial software proposes to automatically detect geometric interfaces between assembly components. However, their algorithms use a global proximity tolerance to find face pairs of components and they don't produce explicitly the geometric model of their contact area. In the assembly simplification method of Russ et al. [35], the authors propose to set components dependencies to remove groups of components having no influence on a simulation or to replace them by defeatured equivalent ones. However, the parent/child relationships created from constraint placement of components do not guarantee to obtain the entire neighborhood of a component because these constraints are not necessarily related to the components' geometric interfaces. In addition, DMUs don't contain components' location constraints when assemblies are complex, as occurring in the automotive and aeronautic industries, to ease design modifications during a PDP. Moreover, part naming identification used in this approach is not sufficient because it locates only individual components contained in the assembly without establishing relations with their adjacent components and their associate geometric model, i.e. replacing a bolted junction by an idealized fastener implies simplifying its nut, its screw as well as the hole connected to them in the tightened components. To provide mesh domains connectivity, Lou [28] and Choudria [10] proposes to identify and re-mesh contact interfaces. Quadros [30] establishes sizing functions to control assembly meshes. However, these methods are used directly on already meshed models, which addresses specific PDP configurations where CAD models are not readily available. Clark [11] detects interfaces in CAD assemblies to create non-manifold models before mesh generation but does not consider the interactions between interfaces and components' shapes transformation processes. In [9], we underline that if a component is simplified or idealized, its shape transformation has an influence on the transformation of its neighbors, e.g. a component idealized as a punctual mass impacts its neighboring component interfaces.

Indeed, to adapt an assembly to FEA requirements, geometric transformations derive from simulation objectives and components' functions are needed to geometrically transform groups of components. It is also critical to structure the geometric models of components so that their shape transformations can preserve the DMU consistency [9]. We propose an approach where a DMU is enriched with geometric interfaces between its components, i.e. contacts and interferences, and functional properties. From this enriched model, simulation objectives can be used to specify geometric operators that can be robustly applied to automate components and interfaces shape transformations during an assembly preparation process.

4 DMU ENRICHMENT WITH FUNCTIONAL INFORMATION

4.1 From a component geometry to its functional designation

The principle of DMU functional enrichment can be summarized with the process flow shown on Fig. 2. Its detailed description can be found in [36]. The focus is placed on this approach because, to the

authors' knowledge, Shahwan's approach can automatically produce a functionally enriched DMU from a pure geometric input model, which is the reliable input data addressed here.

Fig. 2 starts from the STEP [21, 22] representation of an assembly model with the extraction of geometric interfaces between components. Two categories of interfaces are addressed: contacts (surface and line ones) and interferences. The latter originates from the conventional representations of components, e.g. a screw is represented with a threaded part defined as a cylinder whose diameter equals the outer diameter of the thread and its nut is represented with a threaded part defined as a cylinder whose diameter equals an inner diameter of the thread; this produces a geometric interference between these components. At this stage, each component boundary gets modified to incorporate the imprints of its neighboring components through contacts and interferences. The geometric interfaces feed instances of '*conventional interfaces*' (CI) classes structured into a taxonomy, \mathcal{T}_{CI} , that binds geometric and symbolic data, e.g. planar contact, spherical partial contact, cylindrical interference, ... Simultaneously, CIs and assembly components are organized into a CI graph: $CIG(C, I)$ where the components C are nodes and CIs I are arcs.

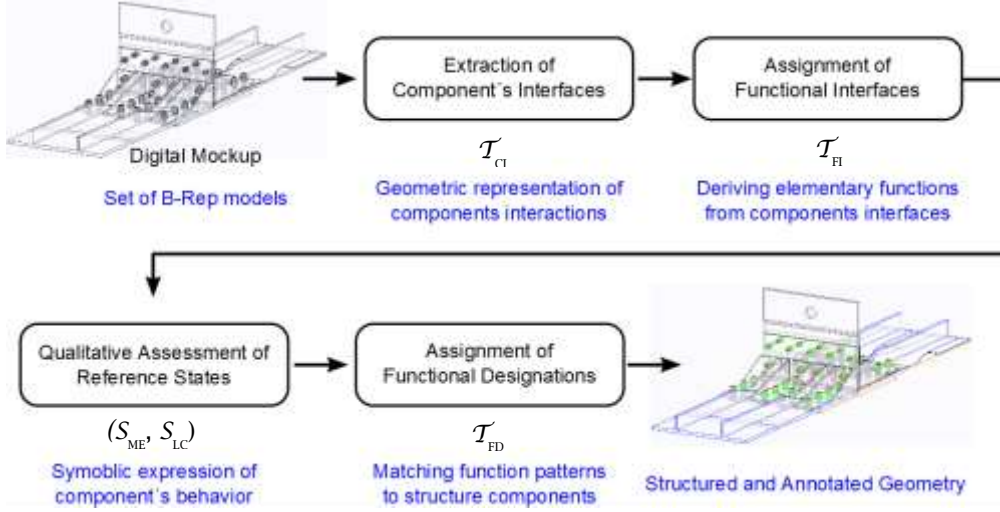


Fig. 2: Process flow to enrich a DMU with functional designations of components and their associated structured geometric models.

Then, instances of functional interfaces (FIs) are derived from CI ones, possibly with multiple instances of FIs originated from a single instance of CI, e.g. a cylindrical contact can become either a cylindrical loose fit or cylindrical tight fit but a planar contact becomes only a planar support. Instances of FIs fit also into a taxonomy \mathcal{T}_{FI} . Now, the duality between geometry and interaction forces is used to derive mechanical parameters, i.e. screws (resulting force, resulting moment), from each instance of FI. These parameters are expressed qualitatively and feed symbolic behaviors of the assembly in order to take advantage of the shape, behavior, function relationship [1, 2, 18, 19]. Assembly behaviors are characterized by reference states and aim at reducing a single FI instance per CI instance. Here, two reference states are of interest: the mechanical equilibrium reference state, S_{ME} , and the static load cycle one, S_{LC} . S_{ME} expresses the fact that each component must hold tight when the assembly has no mechanical interaction with its environment, i.e. each component can satisfy the static equilibrium equations. S_{LC} expresses the fact that any threaded link between any two components of an assembly must generate close loops of forces, i.e. force cycles so that the corresponding components stay at static equilibrium. Fig. 3 illustrates this concept with a bolted junction and a force cycle in magenta.

To this end, it is mandatory to refer to functions. This is achieved with a taxonomy of functions, \mathcal{T}_{FN} , that can produce a functional structure of an assembly (see Fig. 5). Blue items define the sub-path in \mathcal{T}_{FN} hierarchy that characterizes bolted junctions.

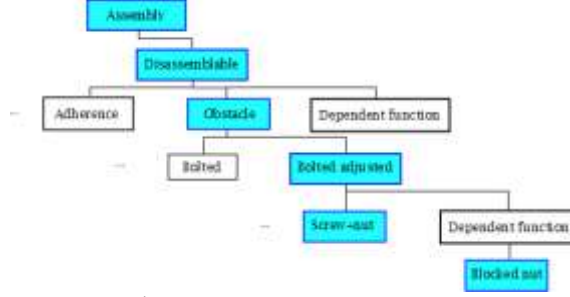


Fig. 5: Subset of \mathcal{T}_{FN} , defining a functional structure of an assembly.

Each instance of a class in \mathcal{T}_{FN} contains a set of components identified by their functional designation, i.e. it contains their structured geometric models and FIs. The generation of these instances takes advantage of the functional designations of components, e.g. adjusted cap screw in the present case, and the load cycles attached to them and derived from S_{LC} . This generation is achieved with inferences on functional designations and properties of load cycles. To this end, it is mandatory to distinguish statically determined configurations from statically undetermined ones. These details are not described here to preserve the paper consistency and concentrate on the exploitation of these functional informations.

As a result of the use of \mathcal{T}_{FN} , a component of a DMU can be automatically identified when it falls into the category of cap screws, nuts, locking nuts that are required to define bolted junctions. This means that their B-Rep model incorporates their geometric interfaces with neighboring components. The graph of interfaces identifies the components a bolted junction contains, the components are assigned functional designations that intrinsically identifies cap screws, nuts, locking nuts, ..., and connects them with an assembly instance in \mathcal{T}_{FN} .

It is now the purpose of the next section to take advantage of this information to set up the template-based transformations.

5 TEMPLATE-BASED TRANSFORMATIONS FOR FE MODELS

As a result, the DMU is now geometrically structured, components are linked by their geometric interfaces, and groups of components can be accurately identified and located in the DMU using their function and geometric structure, e.g. '*adjusted bolted junctions with screw+nut*' (see Fig. 5). Now, the geometric transformations needed to adapt the DMU to FEA objectives are strengthened because screws, nuts, locking nuts can be robustly identified, groups of tightened components are also available through the load cycles attached to cap screws.

Two possible accesses are proposed to define a function-based template related to an assembly function:

- A component through a user-defined selection: from it and its FD, a datastructure gives access to the functions it contributes to. After selecting this component, the user selects the function of interest. Other components are recovered through the selected function this component participates to,
- The function itself that can lead to the set of components needed to define this function and all the instances of this function existing in the targeted assembly.

These accesses can be subjected to constraints that can help identifying the proper set of instances. Constraints aim at filtering out instances when a template is defined from a function to reduce a set of instances to the user's needs, e.g. assembly function with bolts '*constrained with*' tightening plates 'plate_i' and 'plate_j' where these designations are functional designations of these components.

Constraints aim at extending a set of instances when a template is defined from a component, i.e. a single function instance recovered, and needs to be extended, e.g. assembly function with bolts ‘constrained with’ same tightened components and screw head FI of type ‘planar support’ or ‘conical fit’.

5.1 Function-based template and compatibility conditions of transformations

The previous paragraph has sketched how component functions can be used to identify sets of components in an assembly. Indeed, this identification is based on classes appearing in \mathcal{T}_{FN} . Here, the purpose is to define more precisely how the template can be related to \mathcal{T}_{FN} and what constraints are set on shape transformations to preserve the geometric consistency of the components and assembly. Shape transformations are application-dependent and the present context is structural mechanics and FEA to define a range of possible transformations.

The simplest relationship between a template T and \mathcal{T}_{FN} is to relate T to a leaf of \mathcal{T}_{FN} . In this case, T covers instances defining sets of components that contain a variable number of components. T is also dimension independent since it covers any size of component, i.e. it is a parameterized entity. Shape transformations on T are designated as $S_{\mathcal{T}}$ and the template devoted to an application becomes $S_{\mathcal{T}}(T)$. Now, reducing the scope to disassemblable assembly functions and more specifically bolted junctions, one leaf of \mathcal{T}_{FN} can be used to define more precisely T and $S_{\mathcal{T}}(T)$. Conforming to Fig. 5, let us restrict first to the leaf ‘screw+nut’ of \mathcal{T}_{FN} . Then, T contains the following FIs: one threaded link, two or more planar supports (one between nut and plate and at least one between two plates), either one planar support or one conical fit between the screw head and a plate, as many cylindrical loose fits as plates between the screw and plates because the class of junctions is of type adjusted. The shape transformations $S_{\mathcal{T}}(T)$ of T set up to process bolted junctions can be summarized as:

- $S_{\mathcal{T}_1}$: merging screw and nut;
- $S_{\mathcal{T}_2}$: localization of friction effects with a sub domain around a screw;
- $S_{\mathcal{T}_3}$: removal of locking nut if it exists;
- $S_{\mathcal{T}_4}$: screw head transformation for mesh generation purposes;
- $S_{\mathcal{T}_5}$: cylindrical loose fit around the screw shaft to support the contact condition with tightened plates.

Each of these transformations will be detailed throughout the following sections.

Now, the purpose is to define $S_{\mathcal{T}}$ so that $S_{\mathcal{T}}(T)$ exists and preserves the consistency of the components and the assembly. This defines compatibility conditions, C_c , between T and $S_{\mathcal{T}}$ that are conceptually close to attachment constraints of form features on an object [39]. C_c apply to $S_{\mathcal{T}}$ and are introduced briefly here. Given the set of components contained in T , this set can be subdivided into two disjoint subsets as follows:

- I_c is the set of components such that each of its components has all its FIs in T , e.g. the screw belongs to I_c . Consequently, components belonging to I_c are entirely in T ;
- P_c is the set of components such that each of its components has some of its FIs in T , e.g. a plate belongs to P_c . Components belonging to I_c are partially in T .

I_c can be used to define a 3D sub domain of T , T_I defined as the union of all components belonging to I_c . Now, if a transformation $S_{\mathcal{T}}$ takes place in I_c and geometrically lies inside T_I , $S_{\mathcal{T}}(T)$ is valid because it cannot create interferences with other components of the assembly, i.e. C_c is satisfied.

Let us consider some of these transformations to illustrate some C_c . As an objective of FEA, the purpose of the assembly model is to analyze the stress distribution between plates and their interactions with bolts. To this end, the stress distribution around the threaded link between a screw and a nut is not relevant. Therefore, one shape transformation, $S_{\mathcal{T}_1}$ is the removal of the threaded link to merge the screw and the nut (see Fig. 6a). $S_{\mathcal{T}_1}$ is always compatible since the screw and the nut belong to I_c , hence the validity of C_c .

Now, considering another transformation, S_{T_2} , is the localization of friction effects between plates around the screw shaft and the representation of the stress distribution nearby the screw shaft. This is modeled with a circular area centered on the screw axis and a cylindrical sub domain around the screw shaft (see Fig. 6b). Indeed, S_{T_2} , is a domain decomposition [12, 13] taking place in the plates belonging to T. Because the plates belong to P, C_c are not trivial. However, S_{T_2} , takes place inside the plates so they cannot interfere with other components, rather they can interfere with the boundary of the plates or they can interfere between them when several screws are close to each other on the same plate (see Fig. 7). In this case, C_c can be simply expressed, in a first place, as a non interference constraint.

Other shape transformations will be listed when describing one example template in section 4.4.

5.2 Shape transformations and function dependency

The previous section has connected T to \mathcal{T}_{FN} in the simplest way possible, i.e. using a leaf that characterizes a single function. The purpose of this section is to analyze into which extent T can connect to classes of \mathcal{T}_{FN} that perform several functions in the assembly.

In a first place, let us review shortly some concepts of functional analysis [38]. There, it is often referred to several categories of functions that are related to a design process, i.e. external, internal, auxiliary, ... However, this does not convey consistency conditions among these functions, especially from a geometric point of view. Here, the current content of \mathcal{T}_{FN} refers to internal functions, i.e. functions strictly performed by components of the assembly. The ‘screw+nut’ function, as part of bolted junctions, is one of them. Bolted junctions can contain other functions. Let us consider the locking function, i.e. the bolt is locked to avoid any loss of tension in the screw when the components are subjected to vibrations. The locking process can take place either on the screw or on the nut. For the purpose of the analysis, we consider here a locking process on the nut, using a locking nut (see Fig. 9a). In functional analysis, this function is designated as auxiliary function but this concept does not characterize geometric properties of these functions.

From a geometric point of view, it can be observed that FIs of the screw, nut and locking nut are located in 3D such that the FI (planar support) between the nut and locking nut cannot exist if the nut does not tighten the plates. Consequently, the locking function cannot exist if the tightening function does not exist. Rather than using the designation of auxiliary function, which is geometrically imprecise, it is referred to *dependent function*.

The concept of dependent functions is inserted in \mathcal{T}_{FN} at different levels of \mathcal{T}_{FN} to attach the corresponding functions when they exist (see Fig. 5). Based on the concept of dependent function, it is possible to extend the connection rule between T and \mathcal{T}_{FN} . Rather than connections at the leaf level, higher level classes can be connected to T if the dependent functions are taken into account in the C_c of shape transformations S_T so that $S_T(T)$ exists and preserves the consistency of the assembly. As an illustration, let us consider T connected to ‘Bolted adjusted’ (see Fig. 5). Now, S_T can cover the class of bolted junctions with locking nut. Let S_{T_3} , be the transformation that removes the locking nut of a bolted junction, which meets also the FEA objectives mentioned earlier. Because S_{T_3} , applies to a dependent function of ‘screw+nut’, it satisfies the C_c and the resulting model has a consistent layout of FIs. Consequently, T can be effectively connected to ‘Bolted adjusted’, which is a generalization of T.

5.3 Template generation

T is generated on the basis of the components involved in its associated function in \mathcal{T}_{FN} . T incorporates the objectives of the FEA to specify S_T . Here, S_T covers all the transformations described previously, i.e. S_{T_1} , S_{T_2} , S_{T_3} . Fig. 6 and 7 illustrates the key elements of these shape transformations.

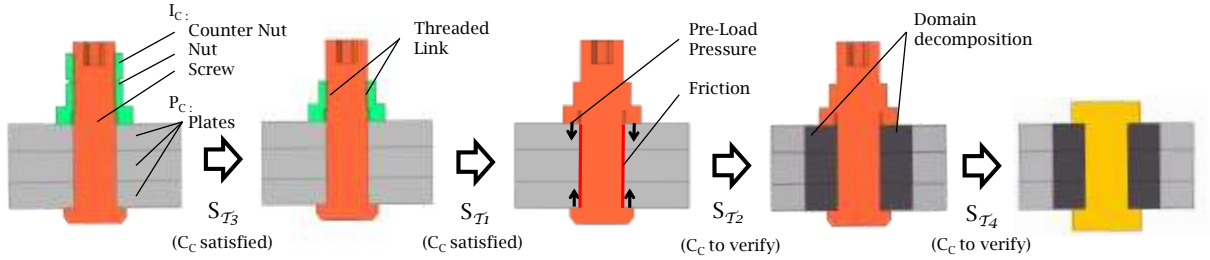


Fig. 6: Compatibility conditions (C_c) of shape transformations S_τ applied to T .

Other shape transformations, $S_{\tau4}$, can be defined to cover screw head transformations and extend the range of screws to flat head ones.

Then, the set of shape transformations structures the dialog with the user to allow him, resp. her, to select some of these transformations. However, the user settings are applied to instances whenever possible, i.e. when the instance belongs to a class where the shape transformations are applicable.

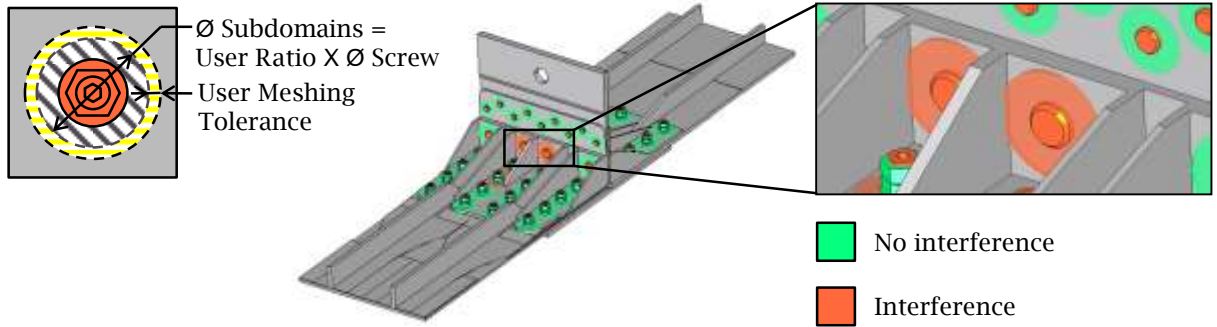


Fig. 7: Checking the compatibility of $S_\tau(T)$ with respect to the surrounding geometry of T .

5.4 Example of template-based operator of bolted junctions transformation

During the PDP of an aeronautical company, simulation engineers perform specific FEAs on assembly sub structures such as the aircraft junction between wings and fuselage. Based on pre-existing physical testing performed by ROMMA project partners, this structure can be subjected to tensile and compressive forces to analyze:

- The distribution of the load transfer among the bolted junctions;
- The admissible extreme loads throughout this structure.

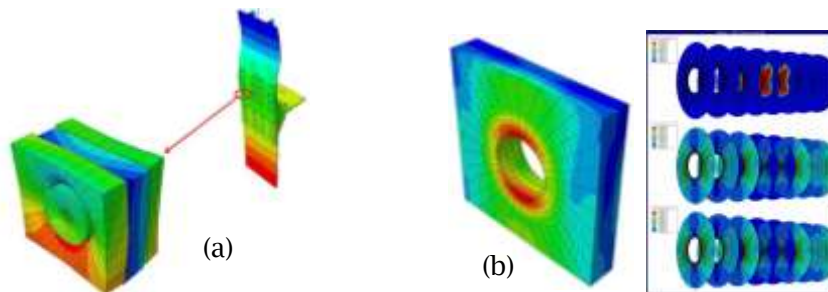


Fig. 8: (a) Multi-scale simulation with domain decomposition around bolted junctions (b) Load transfers at critical holes (courtesy of ROMMA project [4]).

From the physical testing and preliminary numerical models, the following simulation objectives have been set up that initiate the requirements for the proposed template-based transformations.

To adapt the FE model to these simulation objectives while representing the physical behavior of the structure, an efficient domain decomposition approach [12, 13] uses a coarse mesh far enough from the bolted junctions and a specific sub domain around each bolted junction with friction and preload phenomena (see Fig.8a, b). The objective is not to generate a detailed stress distribution everywhere in this assembly but to observe the load distribution areas among bolts using the mechanical models set in the sub domains. The objective of this section is to validate the template-based approach based on a demonstrator that transforms automatically the bolts into simplified sub domains ready for meshing with friction area definition while preserving the consistency of the assembly. The template features are aligned with the needs for setting up a simulation model able to exhibit some of the physical phenomena observed during testing and expressed in the above simulations results.

5.4.1 Operator description

Having enriched the assembly with functional information, the template interface lets the user select a node of \mathcal{T}_{FN} that is compatible with T. In this example, the function to select is: ‘assembly with Bolted junction’ (see Fig. 10). Now, several S_{τ} are either pre set or user accessible. S_{τ_3} is user accessible and set here to remove the dependent function ‘locking with locking nut’ because the simulation objective does not focus on dynamic effects. Then, S_{τ_5} is pre set in T to preserve the cylindrical loose fit between screw and plates to set up contact friction BCs without interpenetration over these FIs. S_{τ_1} is also pre set as well as S_{τ_4} . The latter is performed on the FI (planar support) between the screw head/nut and the plates to obtain a meshing process independent of nut and screw head shapes. Now, T can cover any bolted junction to merge screw, nut and locking-nut into a single domain, reduce the screw and nut shapes to a simple shape of revolution while preserving the consistency of its interfaces. To precisely monitor the stress distribution around bolts and friction between plates, S_{τ_2} is user-selected. It is a simplified model of the Rotscher’s cone [8] (see Fig.9) that enables generating a simple mesh pattern around bolts.

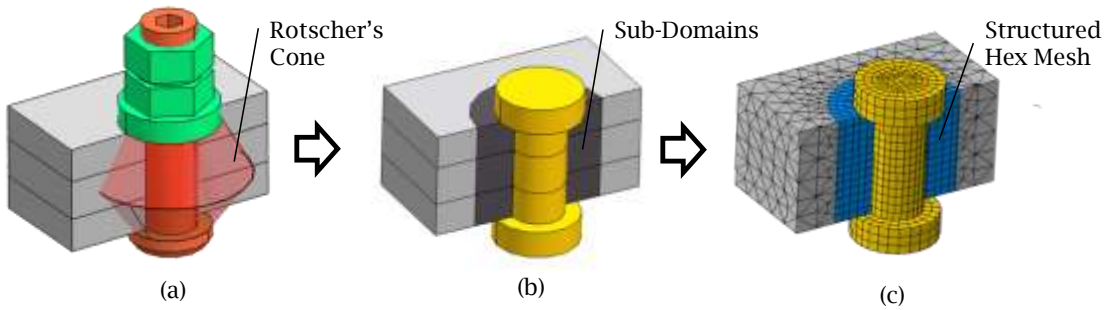


Fig. 9: Template based transformation $S_{\tau}(T)$ of a bolted junction into simple mesh model with friction and contact areas definition around screw and nut.

Based on T, $S_{\tau}(T)$ is fairly generic and parameterized to intelligently select and transform bolts, i.e. it is independent of the number and thicknesses of plates, of the screw diameter, the length and head type (cylindrical (see Fig. 9a) versus flat ones) in addition to the location of each bolt.

Here, $S_{\tau}(T)$ contains S_{τ_2} , a generation of sub domains taking into account the physical effects of the Rotscher’s cone. This geometric transformation could interact with plate boundaries to change the shape of these sub domains and influence the mesh generation process. Presently, templates are standalone entities and are not taking into account these effects left for future developments. At present, the user can adjust the sub domain to avoid these interactions (see Fig. 10a).

5.4.2 Implementation and results

The developed prototype is based on OpenCascade and Python scripting language. The DMU is imported as STEP assembly models, the geometric interfaces between components are represented as independent trimmed CAD faces with identifiers of the initial face pairs of the FI. The assembly functional description is imported as a text file from the specific application performing the functional enrichment described in [36] and linked to the assembly model by components' identifiers.

Fig.10a shows the user interface of the prototype. When selecting the 'assembly with Bolted Junction' function, the user has a direct access to the list of bolted junctions in the assembly. To allow the user filtering his selection, DMU parameters are extracted from the FD of components, e.g. the screw and nut type, the number of tightened components, or from geometry processing based on FDs, e.g. screw diameter. Using these parameters, the user is able to select bolted junctions with diameter between 10 and 16 mm (see Fig.10b) or bolted junctions with screw and locking nut (see Fig.10c), etc. The user can monitor the Rotscher's cone dimension with a FEA parameter called 'sub domain ratio' that represents the ratio between the screw nominal diameter and the sub-domain diameter (see Fig.10d and e). Then, the user-defined 'meshing tolerance' is used during the verification phase to check the compatibility conditions, C_c , between instances and their surrounding geometry (see Fig.7).

Fig. 11 shows two results of the template-based transformations on aircraft structures:

1. A junction between the wing and the fuselage. The assembly contains 45 bolted junctions with 3 different diameters and 2 different screw heads;
2. An engine pylon. The assembly contains over 250 bolted junctions with identical screws and nuts.

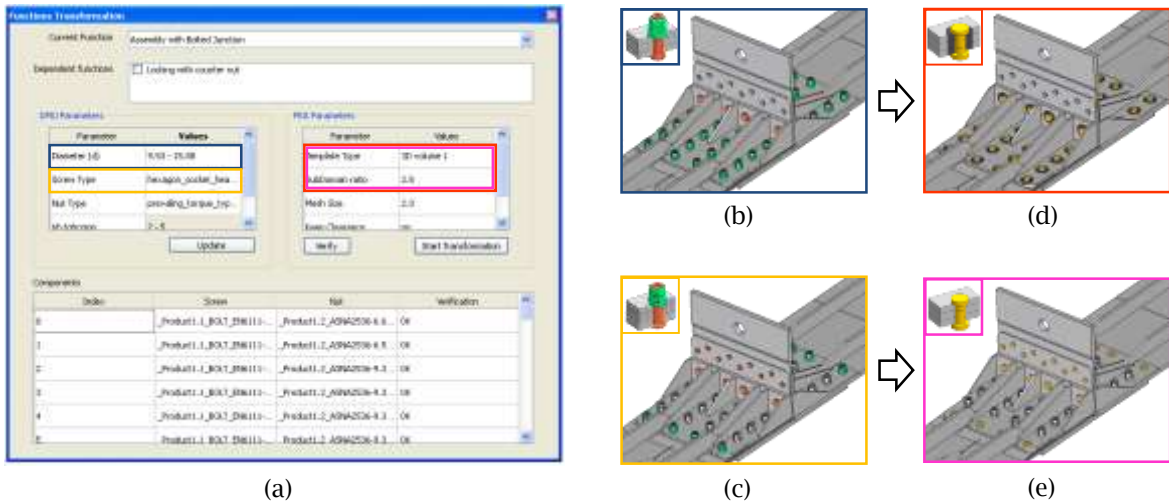


Fig. 10: User interface for transformation of 'assembly Bolted Junctions' (a), filtering bolts based on diameters (b) or screw type (c), template-based transformations with (d) or without (e) sub domains.

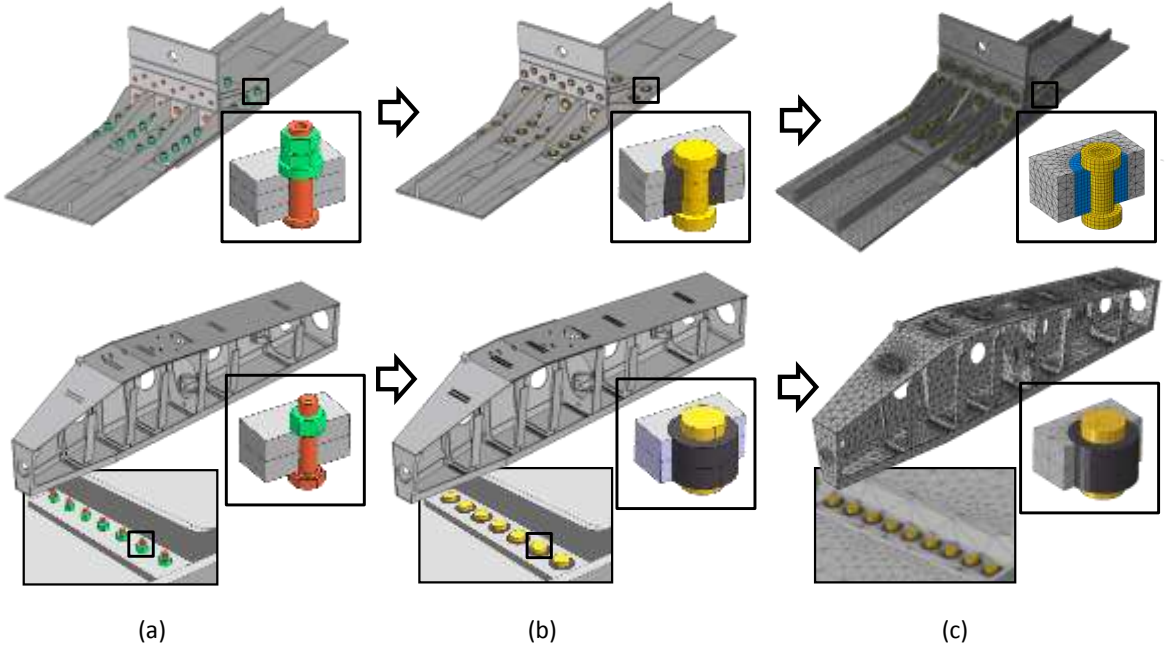


Fig. 11: (a) CAD model with functional designations and geometric interfaces, (b) model after applying $S_r(T)$ on bolts, (c) mesh assembly model with friction area definition.

The final CAD assembly (see Fig. 11b) with simplified bolted junctions has been exported to a CAE software, i.e. Abaqus. STEP files transfer the geometric model and associated xml files describes the interfaces between components to trigger meshing strategies with friction area definition.

Comparing with the process pipeline used with existing industrial software, the figures are as follows. The model preparation from CAD software to Abaqus simulation model takes 5 days of interactive work for aircraft structure 1 mentioned above. Using the pipeline performing the functional enrichment of the DMU and the proposed template-based shape transformations, to directly produce the meshable model in Abaqus and perform the mesh in Abaqus, the overall time is reduced to one hour. The adequacy of this model conforms to the preliminary numerical models set up in ROMMA project and extending this conformity to testing results is ongoing since the template enables easy adjustments of the mesh model. Regarding the aircraft structure 2, there is no reference evaluation of its model preparation time from CAD to mesh generation because it is considered as too complex to fit into the industrial product development process. However, it is possible to estimate the time reduction since the interactive time can be linearly scaled according with the number of bolted junction. This ends up with 25 days of interactive work compared to 1.5 hour with the proposed approach where the time is mostly devoted to the mesh generation phase rather than the template-based transformations, which stay automated.

Though the automation is very high, the template-based approach leaves the user with meaningful parameters enabling him/her to adapt the shape transformations to subsets of bolted junctions when FE is required models.

6 CONCLUSION

In this work, a template-based method empowers the use of DMUs enriched with geometric interfaces and functional information, to automate CAD assembly pre-processing and generate a FE-friendly equivalence of this assembly. Because the template is directly connected to assembly functions, its instances contain a group of components consistently involved in the same assembly function. When

selecting a specific function, the user has a direct access to the associated components, their structured model as well as their FIs. It has been shown also that the template can cover a rather large range of configurations since it is parameterized with respect to components' dimensions, components' number, components' shapes and finally, component function with the concept of dependent function. Using the template-based transformations, the user can robustly and efficiently define the geometric transformations according to his/her FEA objectives and meshing constraints. Thus, new components' shapes adapted to CAE software requirements are produced while the consistency of the assembly model is preserved. The template can be instantiated over the whole assembly to transform quickly repetitive configurations such as bolted junctions which were highly time consuming with interactive processing. Finally, a demonstrator has been presented to prove the efficiency of the template-based transformation approach and its valuable contribution to a PDP. The template-based approach show how shape transformations highly benefit from functional information and strengthen these transformations in the context of complex models like assemblies where many components interact with each other. Bolted junctions of two industrial use-cases have been automatically transformed into simplified macro elements with sub-domains generation. The resulting model produces a seamless generation of FE models conforming to the simulation objectives while taking into account meshing constraints.

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